ELECTRODYNAMIC DUST SHIELD FOR SOLAR PANELS ON MARS. C.I. Calle<sup>1</sup>, C.R. Buhler<sup>2</sup>, J.G. Mantovani<sup>3</sup>, S. Clements<sup>4</sup>, A. Chen<sup>5</sup>, M.K. Mazumder<sup>6</sup>, A.S. Biris<sup>6</sup>, and A.W. Nowicki<sup>2</sup>. <sup>1</sup>Electrostatics and Materials Physics Laboratory, NASA Kennedy Space Center, YA-C2-T, Kennedy Space Center, FL 32899; <sup>2</sup>ASRC Aerospace, ASRC-15, Kennedy Space Center, FL 32899; <sup>3</sup>Florida Institute of Technology, Melbourne, FL 32901; Department of Physics and Astronomy, Appalachian State University, Boone, SC 28608, Department of Physics, Oklahoma Baptist University, Shawnee, OK 74804, <sup>6</sup>Department of Applied Science, University of Arkansas at Little Rock, Little Rock, AR 72204.

**Introduction:** The Materials Adherence Experiment on the Mars Pathfinder mission measured an obscuration of the solar arrays due to dust deposition at a rate of about 0.28% per day. It was estimated that settling dust may cause degradation in performance of a solar panel of between 22% and 89% over the course of two years [1, 2]. These results were obtained without the presence of a global dust storm.

Several types of adherence forces keep dust particles attached to surfaces. The most widely discussed adherence force is the electrostatic force. Laboratory experiments [3] as well as indirect evidence from the Wheel Abrasion Experiment on Pathfinder [4] indicate that it is very likely that the particles suspended in the Martian atmosphere are electrostatically charged.

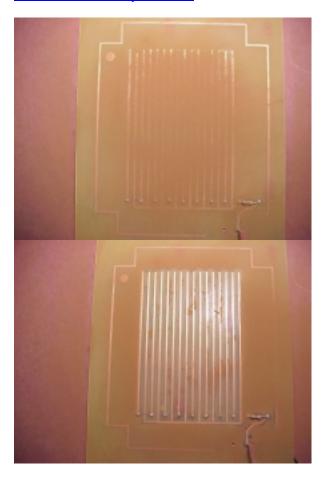
**Electrodynamic Dust Shield:** Masuda and collaborators at the University of Tokyo developed a method to lift and transport particles called the *electric curtain*, in which a series of parallel electrodes connected to an AC source generate a traveling wave that acts as a contactless conveyor [5-8].

We have developed an Electrodynamic Dust Shield prototype to remove dust from surfaces using electrodes that are alternately connected to an AC source and to ground. The electrodes are embedded in a transparent dielectric film to decrease their breakdown potential. Several prototype shields have been tested to check the cleaning efficiency as a function of various line widths and spacings. We used square waveforms in which the frequency was varied from 0 to 500 Hz and the voltage was varied from 0 to 10 kV [9].

Experiments under terrestrial environmental conditions were performed with a screen with 0.7 mm thick electrodes at a spacing of 1.5 mm. The screen was coated with two thin coats of polyurethane in order to prevent spark discharge. A thin coating of the JSC Mars-1 Martian simulant [10] was deposited onto the prototype shild. The simulant dust was baked out at  $150^{\circ}\text{C}$  for several days to remove moisture and the particle sizes were less than 50  $\mu m$  in diameter. About 4000 Volts (peak-to-peak) at 500 Hz was applied to the screen to remove the fine adhesive dust. Dust removal was observed at voltages as low as 800  $V_{p\text{-}p}$  under ambient (Earth) conditions.

Fig. 1 shows the dust-laden Electrodynamic Dust Shield before and after the voltage was applied. Application of the AC field showed that positively charged particles moved to the left while negatively charged particles were deposited to the right of the screen. A short movie of this experiment is located at:

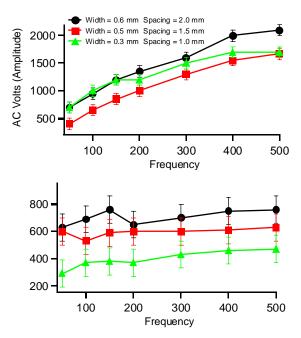
http://physics.ksc.nasa.gov/CurrentResearch/ElectrodynamicScreen/Electrodynamic.htm



**Figure 1. (top)** The dust-laden screen before the voltage was applied and **(bottom)** dust removal afterwards. This experiment was performed at terrestrial environmental conditions.

**Experiments Under Simulated Martian Pressures:** Further experiments using various spacings and

widths are shown below. Figure 2 shows the minimum voltage amplitude (square wave) required for particle transport as a function of frequency for three screens at both atmospheric pressure and simulated Martian pressures of 7 torr CO<sub>2</sub> The voltage values for particle movement at atmospheric pressure are consistent with the literature [8] with lower voltages required for lower frequencies. It is well known that the lower pressure of the Martian environment is conducive to electrical breakdown due to Paschen's Law, which limits the strength of the electric field allowed between two conducting surfaces. Therefore, experiments were designed to check the feasibility of using this approach under Martian conditions. We present the results for the Electrodynamic Dust shield at low pressures in  $CO_2$ .



**Figure 2.** The minimum voltage required for particle transport as a function of frequency for screens of various widths and spacings at (top) atmospheric pressures and (bottom) at simulated Martian pressures 7 torr  $CO_2$ .

At lower pressures (Fig. 2, bottom), the Electrodynamic Dust Shield performed extremely well. The voltages required were very low and the efficiency was not a strong function of the frequency. Much lower voltages were required for transport and the cleaning efficiency was enhanced. Due to the much lower atmospheric pressure on Mars the initial change in momentum of the dust particles is greater. As a result, smaller electrostatic forces (lower electrode voltages) are needed to lift and move these dust particles. The weaker gravity on Mars should also aid in the removal

process. The electrical breakdown of the gas did not occur until the amplitude of the potential exceeded 800 volts (1600  $V_{p-p}$ ) with the three prototype screens.

The dust deposition depended highly on the spacing and width of the screen. For the largest spacing (black line in Fig. 2), the particles deposited in the region between the electrodes, while the smaller spacing (red line) deposited the particles onto the grounded electrodes. Complete removal was not obtained (with this signal-phase signal) until the spacing was down to 1.0 mm (green line) for the given signal. For continuous operation, the prototype screens were  $5 \times 10$  cm in size and required less than 0.02 W/cm<sup>2</sup> when activated. On Mars, this system will be activated only when dust removal is required, perhaps once a sol. Therefore, the daily power requirement will be relatively low. In these experiments, we completely covered the screens with approximately 0.5-1.0 grams of simulant, far more than the expected 30,000 particles per cm<sup>2</sup> [11].

Preliminary results with these unoptimized screens indicate that this technology can be applied under Martian conditions without electrostatic breakdown problems. Furthermore, the low voltages and frequency requirements should not pose serious power, mass or safety concerns for future space vehicles.

References: [1] Landis, G.A., Acta Astronautica, **38**, No. 1, 885 (1996). [2] Landis, G.A. and Jenkins, P.P., JGR, 105, 1855 (2000). [3] Calle C.I., J.G. Mantovani, C.R. Buhler, E.E. Groop, M.G. Buehler and A.W. Nowicki, Proc. of the ESA-IEEE Joint Meeting on Electrostatics, Laplacian Press, Morgan Hill, CA, pp. 90 (2003). [4] Farrel, W.M., M.L. Kaiser, M.D. Desch, J.G. Houser, S.A. Cummer, D.M. Wilt, and G.A Landis, JGR, 104, 3795 (1999). [5] Masuda, S., Advances in Static Electricity, 1, Auxilia, S.A., Brussels, 398 (1970). [6] Masuda, S., Proc. of Albany Conference on Electrostatics, (1971). [7] Masuda, S. Fujibayashi, K., Ishida, K., and Inaba, H., Electronic Engineering in Japan, 92, 9 (1972). [8] Masuda, S., and Matsumoto, Y., Proc. of the 2<sup>nd</sup> International on Static Electrification, Frankfurt (1973). [9] Sims, R.A., Biris, A.S., Wilson, J.D., Yurteri, C.U., Mazumder, M.K., Calle, C.I., and Buhler, C.R., Proc. of the ESA-IEEE Joint Meeting on Electrostatics 2003, Laplacian Press, Morgan Hill, CA, 814 (2003a [10] Allen, C.C., Jager, K., Morris, R., Lindstrom, D., Lindstrom, M., and Lockwood, J., Proc. of the Conference American Society of Civil Engineers, Albuquerque, NM, 469, (1998). [11] Landis, G.A., Acta Astronautica, 38, No. 1, 885 (1996).